

Micromechanical Modelling of Elastic Wave Velocity Variations toward the failure of Brittle Rocks

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Key Points:

- We studied the influence of confining pressure on stress induced anisotropy in crustal rocks.
- We developed a micromechanical model allowing to predict the evolution of elastic velocities toward the failure of brittle rocks.
- The off-fault dissipated energy is consistent with the evolution of the stiffness tensor due to micro-cracks propagation.

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Abstract

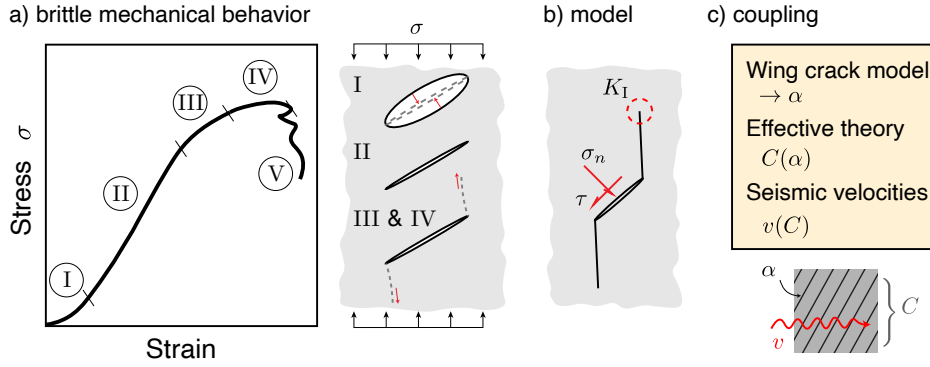
We conducted triaxial experiments on intact specimens of Westerly granite, a proxy for the continental crust. The experiments were performed at confining pressures ranging from 2 to 180 MPa to study the influence of crustal depth on the coalescence of microcracks up to macroscopic failure. Acoustic emissions and elastic wave velocities were monitored throughout the experiments, enabling a comprehensive description of the evolution in anisotropy and crack propagation. In a second step, we developed a micromechanical wing crack model coupled with the cracked solid theory to predict the evolution of elastic wave velocities towards the failure of brittle rocks. The predictions were compared to the experiments conducted at different confining pressures, and a strong correlation between modeled and measured velocities was observed. In addition, our estimates of the energy dissipated in inelastic processes through mechanical measurements is comparable with the energy dissipated in the creation of cracks explaining the measured variations in wave velocities. These results suggest that most of the energy dissipated toward the failure of specimens is related to crack propagation, and that our micromechanical model provides a good physical understanding of the failure of brittle rocks in terms of both damage and elastic wavespeed variations. Moreover, the significance of these inelastic energies indicate that precursory signs of failure might be observed. Therefore, we used our unified model to estimate the expected change in elastic velocities toward the failure of the brittle crust.

1 Introduction

The deformation of crystalline rocks is primarily accommodated by brittle mechanisms at shallow depths (up to 15-30 km depth [Brace and Kohlstedt, 1980]). In this regime, the nucleation, the development, and the coalescence of cracks can lead to the formation of faults, on which catastrophic failures can occur, such as devastating earthquakes. Therefore, understanding the development and the percolation of cracks in brittle materials is of major importance to assess seismic hazard. To this end, multiple theories and models have been proposed to better infer the deformation of the upper crust, as well as the development of damage in materials.

On the first hand, numerous experimental investigations have been conducted to explore the mechanical behavior of rocks leading up to macroscopic fault formation, with a particular focus on crystalline rocks [Brace and Bombolakis, 1963; Lockner, 1995]. Experiments showed that the brittle failure under compression is generally preceded by (figure 1a): i) first an initial closure of existing cracks as compression closes down existing defects, ii) an elastic regime where deformations are reversible, iii) the stable propagation of new cracks in the compression direction when stresses reach critical values on existing defects, iv) followed by their unstable development as their growth make it easier to propagate them further until they coalesce. The development of cracks causes non-linear behavior and create strong anisotropies [Walsh, 1965b,c,d]. v) Finally, localization takes place on a fault where frictional sliding occurs, during which is observed either a rapid release of strain energy, an earthquake in nature, or a slow continuous release of energy in aseismic slip.

On the second hand, these general mechanical observations have been performed through the observation of crack propagation in crystalline rocks. Crack propagation can be monitored through changes in various physical properties, including a change in elastic moduli, elastic wavespeed and rock attenuation, as well as by the onset of dilatancy and acoustic emissions. Specifically, an increase in crack damage has been observed to lead to a decrease in static moduli [Brace and Bombolakis, 1963; Walsh, 1965a]. In addition, seismic velocities are expected to decrease with increasing density of cracks in the material because cracks cause scattering and deflection of seismic waves, leading to longer travel times [Nur and Simmons, 1969; Lockner et al., 1991, 1992; Lockner, 1993]. This scattering of seismic waves on defects generally lead to larger seismic attenuation as the absorption is increased by the number of cracks. Change in attenuation is also related to friction along cracks, which induces dissipation of the energy during the transmission of the elastic waves through inelastic processes activated during the elastic-wave induced strain perturbation [Lockner et al., 1977].



65 **Figure 1.** Framework of the article. a) Typical mechanical curve of a brittle solid exhibiting i) initial clo-
 66 sure of cracks, ii) elastic phase, iii) stable and iv) unstable crack propagation, v) coalescence and frictional
 67 sliding. b) How the wing crack model boils down these mechanisms with linear elastic fracture mechanics
 68 for wings development, and friction on the crack surfaces. c) Proposed coupling of the article; as the model
 69 provides the crack geometry, crack densities α are linked and used to obtain the effective stiffness of the ma-
 70 trix C to estimate seismic velocities v . This coupling allows for continuous recording of brittle mechanisms
 71 toward failure thanks to velocity measurements.

72 To describe and quantify the mechanical behavior of brittle rocks, *Ashby and Sammis* [1990]
 73 developed a simple wing crack model. This micromechanical model considers homogeneously
 74 distributed penny-shaped cracks with friction on their surfaces and with wings that extend un-
 75 til their coalescence. This simplistic representation is particularly adapted to model brittle rocks
 76 because crystalline rocks exhibit these components toward failure at a microscopic level [*Tap-*
 77 *ponnier and Brace*, 1976]. To consider the three main components of matrix behavior, crack open-
 78 ing mechanisms and friction, this model combines different concepts such as (Figure 1b):

- 79 1. Elasticity theory for the matrix with its constants, the Young modulus E and the poisson
 80 ratio ν .
- 81 2. Mohr-Coulomb friction criterion on crack surfaces. The resistance to friction τ under a
 82 normal stress σ_n is:

$$\tau = \mu\sigma_n + c \quad (1)$$

83 where the two model parameters are μ the static friction coefficient, and c the cohesion.

- 84 3. Linear fracture mechanics to describe propagation of wings. Propagation occurs when the
 85 loading increases the stress concentration at the tip of a crack K_I and it reaches a mate-
 86 rial property, the fracture toughness K_{Ic} . So, we note:

$$K_I = K_{Ic} \quad (2)$$

87 These equations govern how the solid acts under any applied loading and how cracks will
 88 propagate. Failure of the rock is considered when the cracks grow enough to interact and finally
 89 coalesce. Due to its simplicity and adequacy, other authors have extended this model to consider
 90 different regimes of crack openings [*Deshpande and Evans*, 2008], the effect of loading rates [*Bhat*
 91 *et al.*, 2012], or subcritical crack growth [*Brantut et al.*, 2012]. Such micromechanical crack mod-
 92 els have only been used for failure predictions. However, they also provide relevant information
 93 regarding the crack geometry and the evolution of damage during loading [*Basista and Gross*,
 94 1998; *Bhat et al.*, 2011; *David et al.*, 2012, 2020a].

95 Importantly, it is also known that the development of damage under differential stress state
 96 induces anisotropy. Because of that, its theoretical effect on effective static and dynamic prop-
 97 erties has been extensively studied [*Horii and Nemat-Nasser*, 1983; *Kachanov*, 1982a,b, 1992;

98 *Sayers and Kachanov, 1995*]. The effective theory of Sayers and Kachanov states that the anisotropy
 99 of rocks due to cracks can be expressed by crack density tensors. These tensors directly reduce
 100 the elastic properties as a function of crack growth and orientation. Therefore, if the cracks' ge-
 101 ometry is known, the crack density tensors can be inverted, which allows evaluating directly the
 102 anisotropic moduli. In short, the effective theory provides a quantification of the anisotropic crack
 103 damage toward failure, plus it allows evaluating seismic velocities in any direction.

104 Despite these advances in both theories, there is currently no unified micromechanical model
 105 allowing for the prediction of the development of elastic anisotropy with wing-crack propaga-
 106 tion in rocks. Indeed, the wing crack model could be linked with the effective theory to provide
 107 estimations of the evolution of elastic wave velocities toward the failure of brittle rocks. If this
 108 coupling is proven to be effective, the model could be used to estimate the evolution of in-situ
 109 stresses by monitoring the evolution of seismic velocities along the fault, excluding possible plas-
 110 tic or healing mechanisms (Figure 1c). Indeed, the analysis of seismic velocities variations has
 111 emerged as a promising tool to assess stress direction and evolution at depth in the crust [*Zoback*
 112 *and Zoback, 1980; Zoback et al., 1987; Zoback and Zoback, 1991; Boness and Zoback, 2006*].
 113 Since the evolution of stress in the Earth's crust is a key parameter controlling the occurrence of
 114 earthquakes, being able to monitor its evolution is of great importance to assess seismic hazard
 115 in a seismogenic area.

116 The goal of this work is to couple and test two of these theories, i.e., micromechanical model
 117 and effective medium theory, to evaluate how accurately they describe brittle mechanisms toward
 118 failure of brittle rocks. In particular, we will use a wing crack model to describe the evolution
 119 of seismic velocities for intact brittle rocks. To this end, laboratory triaxial experiments were con-
 120 ducted at different confining pressures on Westerly granite, a proxy of the continental crust be-
 121 having brittle. Elastic wave velocities were measured with various orientations, allowing to mon-
 122 itor the evolution of the elastic tensor toward the failure of the specimens using effective medium
 123 theory [*Sayers and Kachanov, 1995*]. In a second stage, these experimental results were com-
 124 pared to the predictions obtained using a unified micromechanical model, which allows for the
 125 estimation of the change in elastic properties toward the failure of brittle rocks. We demonstrate
 126 that both experimental and theoretical results are in good agreement, and that our micromechan-
 127 ical model can provide a good estimate of the elastic wave speed in brittle media. Furthermore,
 128 we show that in brittle rocks, most of the energy dissipated during crack propagation is related
 129 to dilatancy, as expected theoretically.

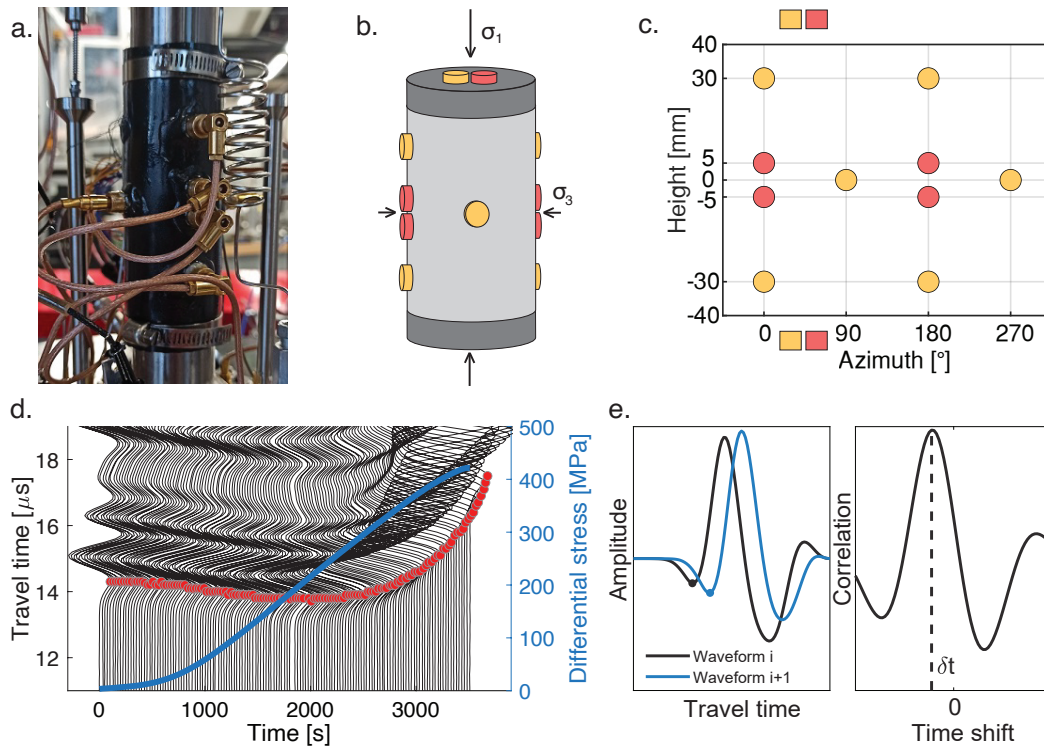
130 2 Methods

131 2.1 Sample and Apparatus

132 Cylinder samples of Westerly granite with a diameter of $d = 38$ [mm] and a height of $h =$
 133 81 [mm] were loaded until failure in the triaxial apparatus called First, installed at the École Poly-
 134 technique Fédérale de Lausanne. An oil confining cell and a compensated axial piston applied
 135 the principal stresses $\sigma_1 > \sigma_2 = \sigma_3$. Five experiments were conducted at confining pressures
 136 of respectively 2, 12, 24, 72 and 180 [MPa]. The samples were protected from oil with a viton
 137 jacket and the pore pressure was null during the experiments. Experiments were conducted by
 138 imposing a constant injection rate of oil in the axial piston chamber, to guarantee a strain rate of
 139 about $\dot{\varepsilon} = 10^{-6}$ [s^{-1}]. Two axial LVDTs and up to eight strain gauges (four radial and four axial)
 140 were used to measure displacements and strains at an acquisition rate of 1 [Hz], respectively.
 141 From the LVDT and gauge measurements, axial and radial strains were averaged (ε_1 and ε_3).

150 2.2 Seismic Velocities

151 Fourteen piezoelectric transducers were placed around the rock sample, following the ar-
 152 rangement presented in Figure 2. Different piezoelectric transducers were used in this study to
 153 monitor both P- and S-wave velocities during experiments. This configuration maximized the
 154 variety of ray path angles. Regularly (around one hundred times per experiment), each transducer



142 **Figure 2.** (a) Picture of the rock assemblage equipped with the sensors, (b) scheme of the sample as-
 143 semblage and (c) sensor map in an azimuthal projection of the sample configuration and arrangement of
 144 piezoelectric sensors. P-waves and S-waves sensors are yellow and red, respectively. (d) Seismic waves evo-
 145 lution during loading (experiment WG5) derived from a pair of P-Waves sensors presenting a raypath angle
 146 of 35° angle). Red dots indicate the arrival time estimated automatically. (e) Cross-correlation method
 147 used to pick the first wave arrival. Waveforms number i and $i + 1$ are resampled for better accuracy, then passed
 148 through a Hann window to limit boundary effects. The similarity is computed, and the maximum correlation
 149 gives the time shift between the two waves.

155 emitted an electric signal while the others recorded reception at 10^7 [Hz]. We obtained ray paths
 156 with the following angles θ : P-waves: 0° , 35° , 52° , 62° , 90° ; S-waves: 90° horizontal and ver-
 157 tical.

158 Thanks to these seismic recordings, arrival times and propagation velocities were computed.
 159 The first arrival times of P and S-waves were initially hand-picked. The following ones were ob-
 160 tained with an iterative cross-correlation method [Brantut *et al.*, 2014] (see Figure 2e). If the cor-
 161 relation coefficient between two waveforms is below 0.9, the arrival time is hand-picked. For im-
 162 proved cross-correlation robustness, first wave peaks are picked. Then, the time interval between
 163 the wave arrival and the peak, which is approximately one-quarter of the wave period, is subtracted
 164 from the travel times. Note that the upper and lower sensors were placed inside the axial steel
 165 of the pistons to avoid stress localization. The travel time of elastic waves through the steel was
 166 subtracted as well. In seismic velocities computations, we assume that the velocities are trans-
 167 versely isotropic. This hypothesis supposedly remains valid until localization of cracks took place,
 168 which was only observed close to failure after peak stress is reached [Lockner, 1993]. Acous-
 169 tic emissions (AE) caused by microseismicity were also completely recorded when five sensors reached
 170 an amplitude threshold. Then, the AE rate was computed with a moving average window of 120
 171 seconds.

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2.3 Seismic attenuation

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Velocity surveys were also used to estimate the evolution of the seismic attenuation towards the failure of brittle rocks [Lockner *et al.*, 1977; Paglialunga *et al.*, 2021]. The attenuation was derived from the waveforms received on each pair of sensors corresponding to a given ray path angle θ . The amplitude reduction of the first wave arrival A is used as a proxy to study attenuation. Only P-wave attenuation is studied because S-wave first arrivals overlap with the P-wave seismic coda. To account for amplitude variations in each sensor, the measurements are normalized by their respective values under hydrostatic pressure as $A/A_h(\theta)$.

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3 Experimental Results

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3.1 Mechanical behavior

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On the five experiments performed on Westerly granite, macroscopic failure is reached shortly after a peak in differential stress. After this peak, the deformations localize on a plane, i.e., fault formation, along which frictional sliding occurs until complete rupture of the sample and nearly total release of differential stress. Peak differential stress strongly increases with confining pressure, going from 199 MPa at 2 MPa confining pressure, to 1081 MPa at 180 MPa confining pressure (table 1). Moreover, the mechanical strain-stress curves document the processes towards failure of the samples. All mechanical curves on figure 3 exhibit similar features: first, they display an initially linear stage, where strains are reversible. On this so-called elastic region, the slope defines the elastic moduli of the material. The slope of $(\sigma_1 - \sigma_3) : \varepsilon_1$, the Young's modulus E_0 , increases with confining pressure between 58 and 74 GPa. Meanwhile, the ratio of transverse and axial strains, the Poisson ratio ν_0 , remains constant at an average of 0.30. These measurements are recapitulated in the table 1. Second, at approximately 70% of the peak differential stress, stress-strain curves deviate from linearity as more strains are accommodated. It coincides with the first acoustic emissions and the onset of dilatancy C' . This point C' is characterized by the maximum positive volumetric strain, and it also increases with increasing confining pressure. The reasons for this change to nonlinear mechanisms will be developed later, but crack propagation admittedly causes it. Initially a stable process, crack propagation becomes unstable as the spike in AE rate and the steep increase in strains both show, until failure. Finally, note that the axial strains towards failure are higher at high confining pressures, but the radial and volumetric strains do not show clear tendencies.

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3.2 Evolution of the elastic wavespeed toward the brittle failure of specimens

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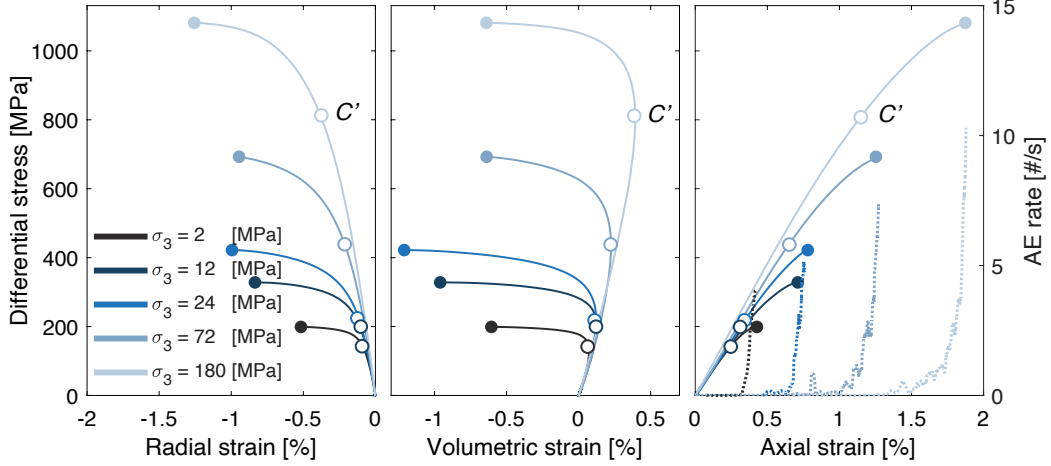
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The evolution of seismic velocities for all the different ray path angles is presented in Figure 4a. As seismic velocities are reduced by cracks intersecting the ray paths, their evolution is a proxy for crack propagation and orientation. We expect velocities perpendicular to the direction of crack growth to get reduced. Before applying differential stress, seismic velocities are nearly isotropic. The average initial v_p are $\sim 4470, 4750, 4990, 5190, 5390$ [m/s] and v_s are 2560, 2790, 2940, 2980, 3050 [m/s] at $\sigma_3 = 2, 12, 24, 72$ and 180 [MPa] respectively; it increases of up to 20% with confining pressure. Then, when differential stress is applied, velocities with ray path angles close to 0° with respect to σ_1 increase of up to 500 [m/s]. However, this effect is less noticeable with high confining pressure with changes of less than 200 [m/s], as observed for $\sigma_3 = 72$ and 180 [MPa]. These increases of velocities are also gradual throughout the loading. Meanwhile, velocities with ray path angles close to 90° stay initially unchanged on all experiments. Upon reaching 50 to 70% of the peak differential stress, elastic wave speeds are reduced until rupture. Velocities are not reduced isotropically; with ray paths approaching a 90° angle, the velocity reductions are down to -25 to -35% depending on the experiment, while for 0° there is generally no reduction. This indicates that cracks mainly grow vertically in the direction of the principal stress. Variations to this rule are also discernible between tests. For instance, on $\sigma_3 = 12$ and 24 [MPa], velocity variations are larger and reach -35% . In these tests, $v_{p,62^\circ}$ decreases more than $v_{p,90^\circ}$ close to failure, indicating possible diagonal coalescence of cracks. After fail-



202 **Figure 3.** Evolution of (a) Radial, (b) volumetric and (c) axial strain with the differential stress for each
 203 experiment. The open symbol corresponds to the onset of dilation (C'). The full symbol; corresponds to the
 204 stress at which macroscopic failure occur. The evolution of acoustic emission rate is presented in panel c.
 205 The elastic phase (straight part of the curve) is followed by a stable crack propagation regime (first AE and
 206 softening of the curve) until unstable propagation happens (AE spike).

226 ure, most of the piezoelectric sensors detached, so velocity measurements are incomplete and not
 227 presented.

233 3.3 Estimate of the evolution of the crack densities

234 Crack densities are inverted from seismic velocities according to *Sayers and Kachanov* [1995].
 235 Elastic wave velocities are directly correlated with elastic properties of rocks and their anisotropy.
 236 For a transversely isotropic medium, the P and S-wave velocities can be calculated in function
 237 of the ray path angle ϕ and the stiffness matrix C [*Brantut et al.*, 2011]:

$$\begin{aligned} v_p(\phi) &= [(C_{11} \sin^2 \phi + C_{33} \cos^2 \phi + C_{44} + \sqrt{M}) / (2\rho)]^{1/2} \\ v_{s,v}(\phi) &= [(C_{11} \sin^2 \phi + C_{33} \cos^2 \phi + C_{44} - \sqrt{M}) / (2\rho)]^{1/2} \\ v_{s,h}(\phi) &= [(C_{66} \sin^2 \phi + C_{44} \cos^2 \phi) / \rho]^{1/2} \end{aligned} \quad (3)$$

238 with the $\rho = 2650$ [kg/m³] density of the medium and M defined as:

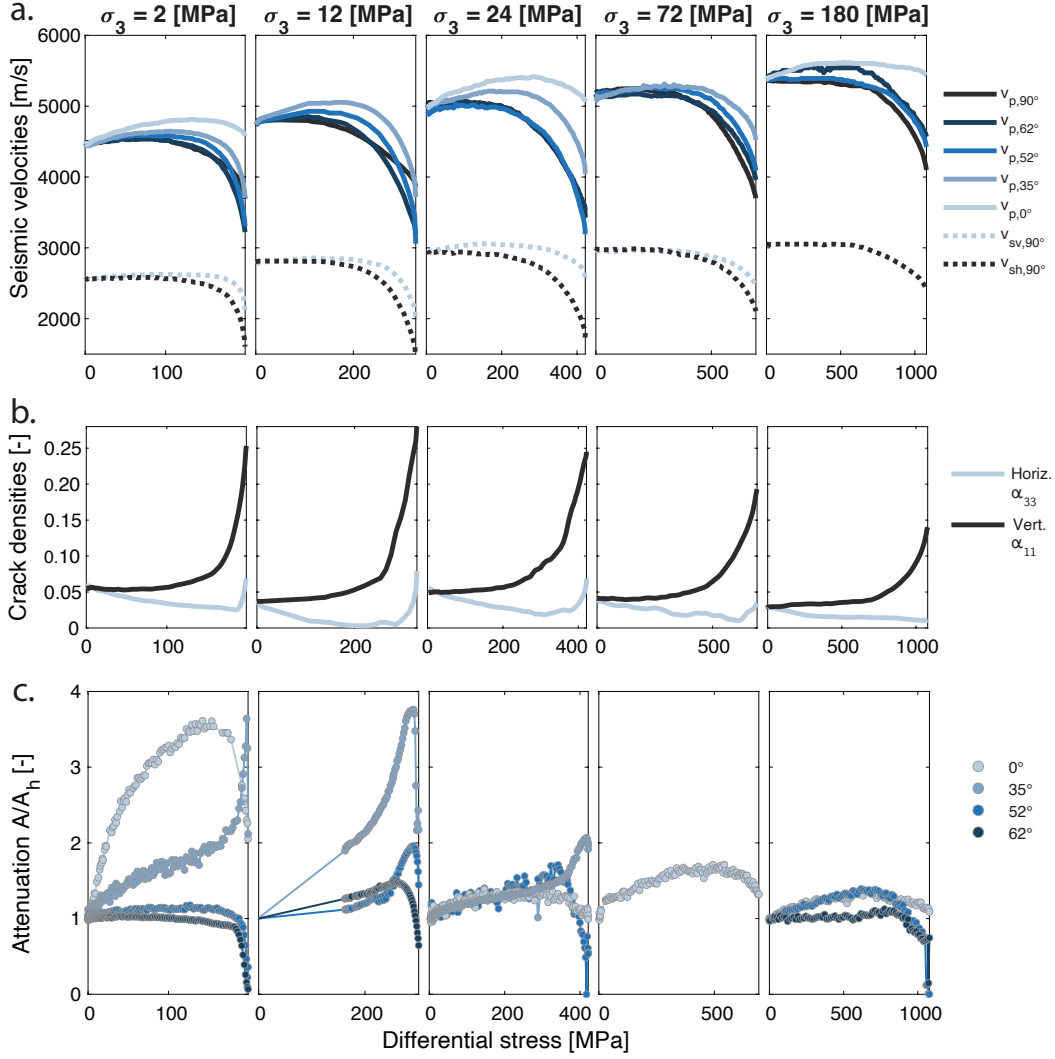
$$M = ((C_{11} - C_{44}) \sin^2 \phi - (C_{33} - C_{44}) \cos^2 \phi)^2 + ((C_{13} + C_{44}) \sin 2\phi)^2 \quad (4)$$

239 The stiffness is a function of the elastic properties and the crack properties that *Sayers and Kachanov*
 240 [1995] describe with the second rank tensor of crack density α . For transversely isotropic cracks,
 241 they obtained:

$$\begin{aligned} C_{11} + C_{12} &= [(1/E_0) + \alpha_{33}] / D \\ C_{11} - C_{12} &= 1 / [(1 + \nu_0) / E_0 + \alpha_{11}] \\ C_{33} &= [(1 - \nu_0) / E_0 + \alpha_{11}] / D \\ C_{44} &= 1 / [2(1 + \nu_0) / E_0 + \alpha_{11} + \alpha_{33}] \\ C_{13} &= -(\nu_0 / E_0) / D \\ C_{66} &= 1 / [(2(1 + \nu_0) / E_0 + 2\alpha_{11})] \end{aligned} \quad (5)$$

242 with:

$$D = (1/E_0 + \alpha_{33})((1 - \nu_0) / E_0 + \alpha_{11}) - 2(\nu_0 / E_0)^2 \quad (6)$$



228 **Figure 4.** Evolution of a) seismic velocities, b) principal crack density components, and c) attenuation of
 229 the amplitude of the first seismic wave arrival. Confining pressure closes existing cracks, which increases
 230 hydrostatic velocities, decreases initial crack densities, and makes wave amplitude variations less sensitive.
 231 They all show that during each experiment, horizontal cracks (perpendicular to differential stress) get initially
 232 closed, then new cracks grow vertically (parallel to differential stress) exponentially until sample failure.

243 The second rank crack density tensor α is defined by:

$$\alpha = \frac{1}{V} \sum_{m=1}^N (a^3 \mathbf{n} \times \mathbf{n})^{(m)} \quad (7)$$

244 For N circular cracks of respective radius $a^{(m)}$ in a volume V , where $\mathbf{n}^{(m)}$ is the unit normal
 245 to each crack. Assuming transversely isotropic damage, the principal values of this tensor are the
 246 vertical crack density α_{11} and the horizontal crack density α_{33} . The pair of values maximizing
 247 the likelihood between observed and theoretical speeds are kept as the components of the measured
 248 crack density.

249 The evolution of the principal components of crack densities are presented in Figure 4b.
 250 These curves allow for an easier and more detailed interpretation of crack propagation than seis-

mic velocities. Initially, the crack distribution is nearly isotropic and the initial values of crack density decreases with increasing confining pressure, from 0.06 to 0.03 at $\sigma_3 = 2$ [MPa] and $\sigma_3 = 180$ [MPa], respectively. As the differential stress slowly increases, we observe an initial closure of horizontal cracks while the vertical crack densities remain unchanged. At high confining pressure, the horizontal crack density varies less, indicating that existing cracks are already closed. After a critical strength is reached during loading, crack nucleation begins and the vertical crack density grows exponentially until crack dilatancy leads to failure. This essentially proves that the cracks extend vertically in the direction of the principal stress σ_1 . Although the number of experiments is limited, the peak crack densities seem to decrease with an increasing confining pressure, likely due to more localized damage at higher stresses. Finally, late horizontal crack opening is observed because of imminent failure.

Table 1. Table of the main results of the experiments

Test name	Confining pressure σ_3 [MPa]	Peak differential stress q_{peak} [MPa]	Young Modulus E_0 [GPa]	Poisson ratio ν_0 [-]
WG4	2	199	57.7	0.31
WG2	12	328	58.9	0.28
WG5	24	422	65.5	0.33
WG6	72	693	70.0	0.30
WG3	180	1081	73.7	0.28

3.4 Evolution of Attenuation During the Experiments

The evolution of attenuation throughout the experiments is analyzed for the different regimes identified before failure (i.e., initial closure of microcracks, elastic loading, and microfracturing).

The increase of differential loading (occurring in this configuration along the vertical direction) forces the closure of mainly horizontal cracks (Figure 4c, note that the experiment at $\sigma_3 = 12$ [MPa] has been replicated for improved attenuation data). Changes in horizontal crack density α_{33} are most likely to affect the direction parallel to the compression axis. This is confirmed by the seismic monitoring along the vertical direction; $A/A_h(0^\circ)$ increases concurrently with a decrease of α_{33} in all the experiments. However, a clear dependence on the applied confining pressure σ_3 is observed, with peak values of $\sim 3.57, 1.34, 1.71, 1.31$ for $\sigma_3 = 2, 24, 72, 180$ [MPa], respectively.

The other directions would also be affected by the closure of horizontal cracks, in proportion to their orientation. $A/A_h(35^\circ)$ reveals itself to be highly affected by the decrease of α_{33} , notably increasing for small applied differential loads. As for the amplitude evolution in the vertical direction ($A/A_h(0^\circ)$), $A/A_h(35^\circ)$ increases during the closure of horizontal cracks and decreases with applied confining pressures. $A/A_h(52^\circ)$ and $A/A_h(62^\circ)$ show a slight increase during the closure of horizontal cracks. This behavior is similar for all the different applied σ_3 . While for low σ_3 (in particular for $\sigma_3 = 2, 12$ [MPa]), a clear distinction between monitoring directions is observable, for higher σ_3 this distinction becomes less evident. For high σ_3 , the increase in $A/A_h(\theta)$ is similar for all directions and reaches peak values of $\sim 1.3, 1.7, 1.3$ for respectively $\sigma_3 = 24, 72, 180$ [MPa], comparable to the values reached by $A/A_h(52^\circ)$ and $A/A_h(62^\circ)$ at low σ_3 .

Then, in the purely elastic loading, attenuation remains essentially unchanged until crack nucleation occurs.

Once the critical strength of the sample is reached, crack nucleation begins and microcracks parallel to the compression axis start to grow (Figure 4c). The increase in the vertical crack

289 density α_{11} is expected to mostly affect $A/A_h(\theta)$ in the directions perpendicular to the newly
 290 created cracks. This trend is confirmed for $A/A_h(52^\circ)$ and $A/A_h(62^\circ)$, which both decrease with
 291 increasing α_{11} for all the tested confining pressures. Unexpectedly, despite the increase of ver-
 292 tical cracks, $A/A_h(35^\circ)$ keeps increasing up to the sample's proximity to failure (this behavior
 293 is, however, compatible with previous observations of compression tests [Lockner *et al.*, 1977]).
 294 Finally, the evolution of $A/A_h(0^\circ)$ shows a decrease with increasing α_{11} , similar to the one ob-
 295 served for $A/A_h(52^\circ)$ and $A/A_h(62^\circ)$, but less pronounced in magnitude.

296 4 Modelling of the Experimental Results

297 4.1 Brittle Mechanisms

298 This experimental study provides a complete record of the influence of the confining pres-
 299 sure on the evolution of stresses, strains, and seismic velocities toward the failure of crystalline
 300 rocks. This data set is now used to calibrate a micromechanical model based on wing-crack the-
 301 ory from *Ashby and Sammis* [1990], coupled with effective medium theory, to attempt to predict
 302 both the brittle strain-stress behavior recorded during experiments and the evolution of seismic
 303 velocities during loading.

304 The rock sample is simplified to an elastic medium with homogeneously distributed cracks
 305 of identical geometry. The sample is submitted to a triaxial loading defined by the principal stresses
 306 σ_1 and $\sigma_2 = \sigma_3$, which impose a shear stress τ and normal stress σ_n on shear crack interfaces.
 307 Sliding along interfaces is assumed when the state of stress reaches a classical Mohr-Coulomb
 308 criterion, defined by a static friction coefficient μ and no cohesion. The geometry of cracks is sim-
 309 plified to N_V identical penny-shaped cracks of radius a , angle Ψ , and two wings of length ℓ as
 310 shown in Figure 5a, b. Since the granite is initially intact, the wings have no initial length ($\ell =$
 311 0) and the number of cracks can be approximated to $N_V = \rho_c/a^3$, where ρ_c is the initial mea-
 312 sured crack density.

313 Geometrically, the stresses acting on the cracks are:

$$\tau = \frac{\sigma_1 - \sigma_3}{2} \sin 2\Psi \quad (8)$$

$$\sigma_n = \frac{\sigma_1 + \sigma_3}{2} + \frac{\sigma_1 - \sigma_3}{2} \cos 2\Psi \quad (9)$$

315 The sum of the horizontal components of these stresses applied on a crack are defined as the wedg-
 316 ing force:

$$F_w = (\tau + \mu\sigma_n)\pi a^2 \sin \Psi \quad (10)$$

317 For an implementation in a three-dimensional setting, *Ashby and Sammis* introduced this expres-
 318 sion:

$$F_w = (A_1\sigma_1 - A_3\sigma_3)a^2 \quad (11)$$

319 Where A_1 and A_3 are:

$$A_1 = \pi \sqrt{\frac{\beta}{3}} (\sqrt{1 - \mu^2} - \mu) \quad (12)$$

$$A_3 = A_1 \frac{\sqrt{1 + \mu^2} + \mu}{\sqrt{1 + \mu^2} - \mu} \quad (13)$$

320 β is chosen to fit the beginning of crack propagation. This wedging force creates a mode I stress
 321 intensity factor, denoted K_I , at the tip of the crack, whose effective length is $\ell + \beta a$:

$$K_I = \frac{F_w}{[\pi(\ell + \beta a)]^{3/2}} + \frac{2}{\pi} (\sigma_3 + \sigma_3^i) \sqrt{\pi \ell} \quad (14)$$

322 If $K_I > K_{Ic}$, the crack opens until $K_I < K_{Ic}$. To consider the effect of crack interaction, σ_3^i
 323 is an added internal stress that equilibrates the wedging force:

$$\sigma_3^i = \frac{F_w}{\Pi - \pi(\ell + a \cos \Psi)} \quad (15)$$

324 With the total crack area projected vertically $\pi(\ell + a \cos \Psi)$, and Π the area per crack:

$$\Pi = \pi^{1/3} \left(\frac{3}{4N_V} \right)^{2/3} \quad (16)$$

325 If $\Pi - \pi(\ell + a \cos \Psi)$ becomes negative, it means that the cracks coalesce and failure is reached.

326 The main parameters of this model are the initial crack density ρ_c , their radius a , the co-
 327 efficient of friction μ , the fracture toughness K_{Ic} and β . We consider cracks with the orientation
 328 $\Psi = 45 + 1/2 \arctan \mu$. In the literature, the fracture toughness of Westerly granite ranges from
 329 1 to 2 [MPa m^{1/2}] [Ashby and Sammis, 1990; Atkinson and Rawlings, 1981; Meredith and Atkin-
 330 son, 1985], so an average value of $K_{Ic} = 1.5$ [MPa m^{1/2}] has been chosen. The size of initial
 331 cracks is estimated to be $a = 0.3$ [mm], which represents half of the average grain size of the
 332 granite. The initial crack densities were computed with seismic velocities recorded under hydro-
 333 static stress conditions; a linear decrease in initial crack densities from 0.16 to 0.09 with increas-
 334 ing confining pressure was observed. The coefficient of friction $\mu = 0.65$ was chosen to fit the
 335 failure envelopes. Finally, β was defined using the onset of the crack opening process in the ex-
 336 periments at each confining pressure tested. We observed a general decrease in the crack open-
 337 ing process with increasing confining pressure, which justifies an increase in β with σ_3 .

338 **Table 2.** Parameters of the wing crack model

Parameter	Value	Remark
K_{Ic}	1.5 [MPa m ^{1/2}]	Average value in the literature
a	0.3 [mm]	Half of grain size
μ	0.65	Slope of the failure envelope
β	0.25 to 0.5	Increasing linearly with confining pressure
ρ_c	0.16 to 0.09	Decreasing linearly with confining pressure

339 The failure envelope estimated from these parameters (table 2) is presented in Figure 5c.
 340 Compared to the Hoek-Brown failure envelope, classically used in geotechnical engineering [Cai,
 341 2010], the accuracy of the wing crack failure envelope is acceptable. The choice of parameters
 342 is critical for the wing crack model, yet each author evaluates them differently. For example, the
 343 size of the initial cracks is a sensitive parameter but hardly measurable; β has different defini-
 344 tions; and the fracture toughness of Westerly granite varies across studies.

352 Parameters of table 2 are now used in the micromechanical model to predict the evolution
 353 of strains measured experimentally. In elasticity, strains are linked to the strain energy density
 354 W :

$$\varepsilon_{ij} = \frac{\partial W}{\partial \sigma_{ij}} \quad (17)$$

355 W can be decomposed between the strain energy of the uncracked solid W_0 , plus the strain en-
 356 ergy of each crack ΔW :

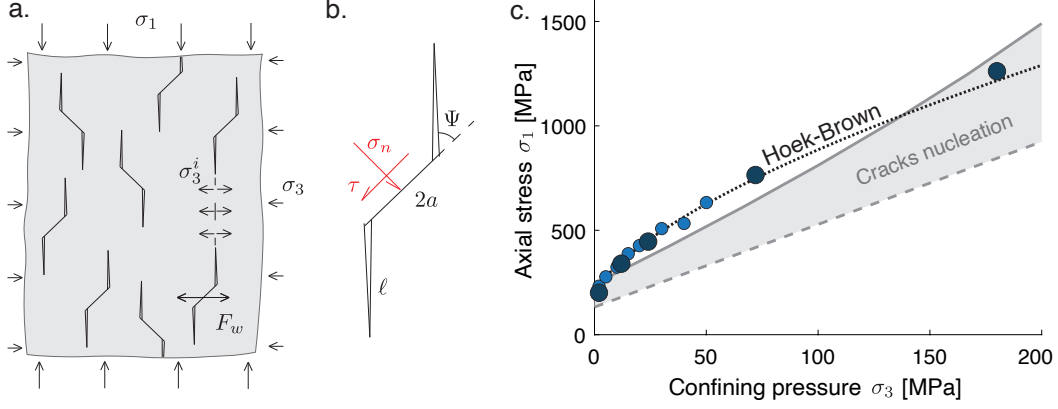
$$W = W_0 + N_V \Delta W \quad (18)$$

357 Where:

$$W_0 = \frac{1}{2E_0} (\sigma_1^2 + 2\sigma_3^2(1 - \nu_0)) \quad (19)$$

$$\Delta W = \frac{1}{E_0} \int_0^a K_I^2 2\pi a da \quad (20)$$

359 K_I is substituted from Eq. 14, which allows evaluating ε_v . For ease of numerical implementa-
 360 tion, the full development of Deshpande and Evans [2008]; Nicolas et al. [2017] is used to com-
 361 pute ΔW . Finally, as $\varepsilon_v = \varepsilon_1 + 2\varepsilon_3$, radial and axial strains are decomposed straightforwardly
 362 as the loading is imposed in axial displacement.



345 **Figure 5.** Geometry adopted for the wing crack model taken from [Brantut et al., 2012; Nicolas et al.,
 346 2017]. a. The rock is simplified to an elastic medium with homogeneously distributed cracks of identical
 347 geometry. b. Cracks are inclined penny shaped, slide on their face and extend vertically at its tips until they
 348 coalesce. The effect of cracks interaction is also considered with an internal stress σ_3^i that equilibrates the
 349 wedging force F_w . c. Experimental results (blue dots) used to fit the failure envelope of the wing crack model
 350 (in gray) and its crack nucleation process highlighted. Hoek and Brown failure envelope (dashed line) is also
 351 presented for comparison purposes.

363 Modelled radial, volumetric and axial strains are presented in figure 6 and compared to ex-
 364 perimental results for each confining pressure tested. As shown, the model reasonably fits the me-
 365 chanical behavior but does not capture accurately the softening of the axial strains as cracks only
 366 grow axially. In addition, experiments exhibited higher radial strains at lower confining pressures.
 367 These differences are probably related to our estimates of the parameters. For instance, differ-
 368 ent authors extended the wing crack model to include creep with subcritical crack growth [Bran-
 369 tut et al., 2012] or different regimes of crack opening [Deshpande and Evans, 2008]. Integrat-
 370 ing these extensions would bring supplementary parameter choices. However, we decided to keep
 371 the simplest form of the model in the following, since the prediction of the mechanical behav-
 372 ior remains close from our experimental observations.

376 4.2 Modelling of the Velocities From Brittle Mechanisms

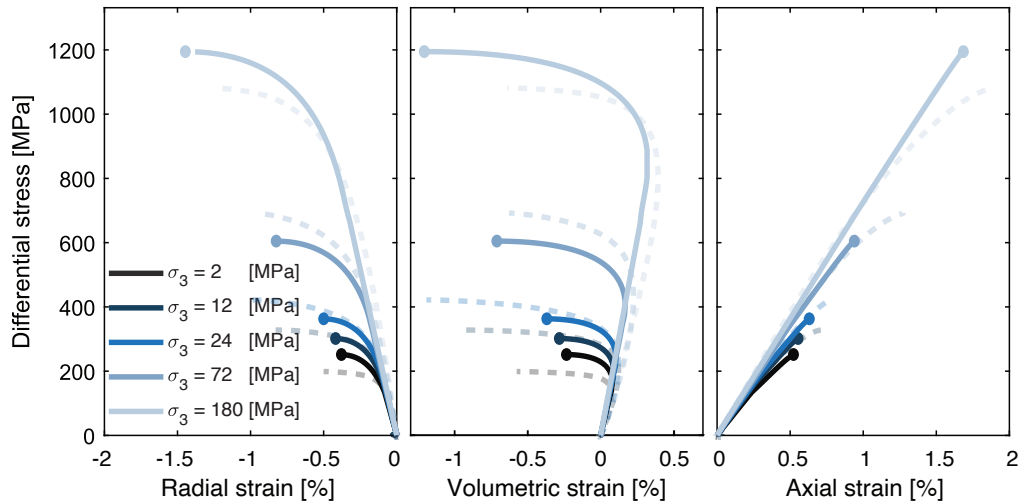
377 To compute velocities from the wing crack model, Kachanov cracked solid theory was used.
 378 The crack density is obtained straightforwardly from its definition: $\rho_c = N_V a_{eq}^3$. As in section
 379 3.3, we assume here a transversely isotropic cracks distribution, so that $\rho_c = 2\alpha_{11} + \alpha_{33}$. A
 380 simplification in the model compared to the experimental case is that the cracks only grow ver-
 381 tically in the wing crack model, while the horizontal crack density component stays unchanged.
 382 This simplification implies that only the vertical crack density increases with the propagation of
 383 wing cracks during the loading in axial stress, following:

$$\begin{aligned}
 \Delta\alpha_{11} &= \Delta\rho_c/2 = N_V(a_{eq}^3 - a^3)/2 \\
 \Delta\alpha_{33} &= 0
 \end{aligned}
 \tag{21}$$

384 where a_{eq} is the radius of an equivalent penny shaped crack composed by both the shear cracks
 385 and the associated wing cracks, defined geometrically by:

$$\begin{aligned}
 a_{eq} &= a \cos(\theta_{eq} - \Psi) + 2l \cos \theta_{eq} \\
 \theta_{eq} &= \arctan \frac{\sin \Psi}{2l/a + \cos \Psi}
 \end{aligned}
 \tag{22}$$

386 In a second step, the theoretical estimates of $\Delta\alpha_{11}$ are used to estimate the evolution of the ve-
 387 locities with increasing differential stress, following the equations described in section 3.3.



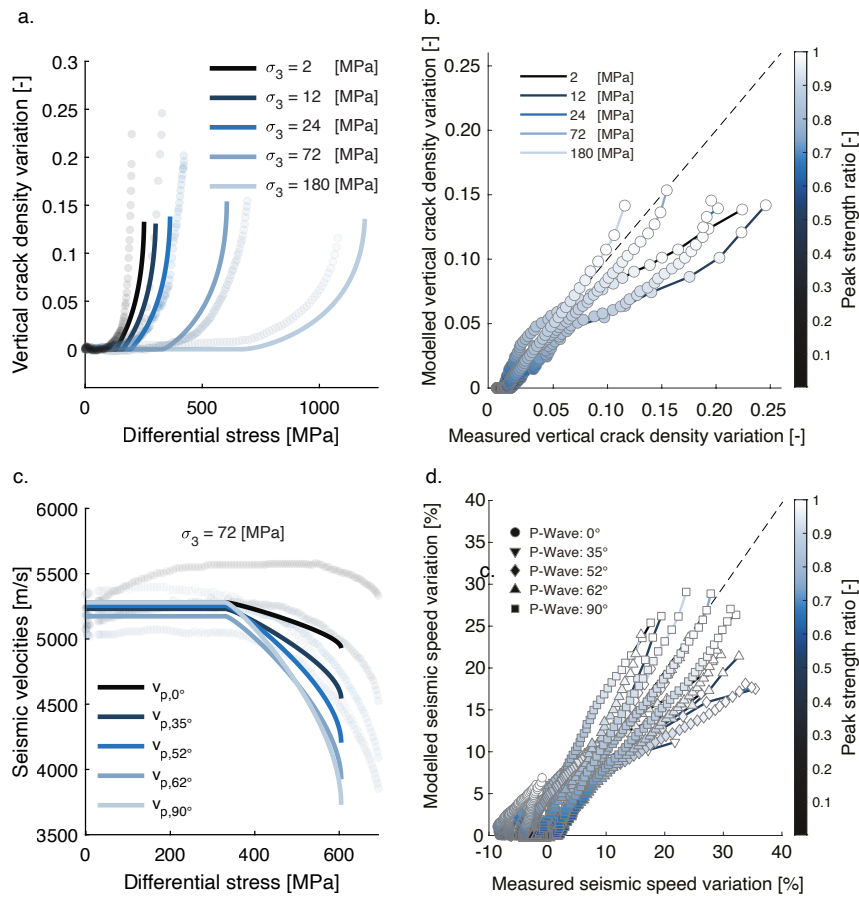
373 **Figure 6.** Radial, volumetric and axial strain stress plots of the model. Dashed lines are the experimen-
 374 tal results and show where the model is inexact: radial strains are higher at low confining pressures and the
 375 model does not consider the propagation of horizontal cracks.

388 4.3 Accuracy of the micromechanical modelling

389 The modelled crack densities and velocities are directly compared with the experimental
 390 results. The figures 7a and b present our predictions of the evolution of $\Delta\alpha_{11}$ toward the failure
 391 of crystalline rocks for each confining pressure tested experimentally. A good correlation is glob-
 392 ally observed between both experimental results and theoretical predictions. However, the theo-
 393 retical estimate of the vertical crack densities are overestimated close to failure at the lower con-
 394 fining pressure tested, (at 2, 12 and 24 [MPa]).

395 Regarding seismic velocities, comparisons are presented in figures 7c and d. The figure 7c
 396 presents specifically the experiment conducted at $\sigma_3 = 72$ [MPa]. Here, $v_{p,35-90^\circ}$ are all mod-
 397 elled accurately with a sharp decrease of up to 30%. However, a mismatch occurs for $v_{p,0^\circ}$ as
 398 the experimental velocities increase of 300 [m/s], while the modelled velocities remain nearly
 399 constant. The figure 7d allows making more general observations to understand these inconsis-
 400 tencies. Similarly to vertical crack densities, the elastic wave velocities generally diverge close
 401 to failure. At this stage, strain localization might occur, explaining the mismatch between exper-
 402 imental results and theoretical ones. In addition, significant deviations are observed when the ray
 403 paths are oriented close to 0° and before the crack nucleation, related to the fact that our model
 404 does not consider the closure and opening of horizontal cracks. Nevertheless, this simplification
 405 remains generally coherent, as the experimental horizontal crack density components have vari-
 406 ations less than 0.05 (figure 4b).

407 The wing crack limitations addressed are: i) even though the wing crack model considers
 408 stress intensity factor at the defects, its failure envelope is similar in many aspects to a Mohr-Coulomb
 409 failure envelope in reason of shear on the crack surfaces. The envelope is close to a straight line,
 410 and most of the further limitations can be reduced to failure predictions uncertainties. ii) The wing
 411 crack model only considers vertical crack propagation, failing to take into account diagonal co-
 412 alescence close to macroscopic failure and initial cracks closure. Seismic velocities can be pre-
 413 dicted with the model, but with prudence when the cracks are not perpendicular to the ray path.



414 **Figure 7.** a. Evolution of the modelled vertical crack density during loading. Experimental results are in
 415 transparency. b. Direct comparison between the vertical crack density estimated for each velocity surveys and
 416 the one predicted using micromechanical model at the similar state of stress. The dashed line presents of slope
 417 of 1. The color bar corresponds to the stress state at which the surveys are performed. c. Modelled evolution
 418 of velocities during loading for the experiment conducted at 72 [MPa] confining pressure. Experimental re-
 419 sults are presented in transparency to compare with the theoretical predictions. d. Direct comparison between
 420 experimental and modelled elastic wave speeds obtained at each velocity surveys. The color bar corresponds
 421 to the stress state at which the surveys are performed. The dashed line presents of slope of 1.

422 5 Discussion

423 This study provides experimental results and a model describing crack-induced anisotropy
 424 of brittle rocks toward their failure. Specifically, we studied the influence of confining pressure,
 425 which is a direct proxy of the depth in the crust. Therefore, the analysis of this full record can
 426 provide new insights of failure mechanisms potentially observable in crustal conditions.

427 We first discuss how confining pressure affects the crack development and induce anisotropic
 428 changes such as observed by seismic attenuation. Then, the validity of the wing crack model is
 429 analyzed through an energy budget that brings: i) a physical validation of the model, ii) details
 430 on which inelastic dissipative processes (i.e., dilatancy, new crack surfaces formation, shear slid-
 431 ing) are predominant in the brittle regime, and iii) comparison with energy dissipated during fail-
 432 ure to discuss whether precursory elements of earthquakes can be observed with seismic veloc-
 433 ity variations. Finally, we extrapolate the results of the model to estimate seismic velocities to-
 434 wards failure or brittle rocks in the crust.

5.1 Influence of confining pressure on crack induced-anisotropy and seismic attenuation toward the failure of crystalline rocks

The mechanical behavior of brittle solids is separated into the following regimes: i) initial closure of existing cracks, ii) purely elastic regime, iii) stable crack propagation, iv) unstable crack propagation, and v) failure. In our experiments (Figure 3), the initial closure due to deviatoric stress is barely detectable, as the confining pressure has already closed the majority of cracks. In the elastic stage, the increase of Young's modulus with confining pressure is explained by grains locking and by remaining crack closure, which both stiffen the rock matrix. Then, the non-linear strain hardening behavior is caused by the nucleation of new cracks and propagation of already existing cracks. Crystal plasticity can be excluded, as observed under these pressures on westerly granite [Brace *et al.*, 1966]. Crack opening is a dilatant process and requires a significant amount of energy and high stresses to open cracks at high confining pressure compared to low confining pressure. It explains why the onset of dilatancy, acoustic emissions, and crack nucleation occurs later at higher confining pressures [Brace *et al.*, 1966; Lockner, 1993; Paterson and Wong, 2005; Browning *et al.*, 2017]. Unstable crack propagation is highlighted by the change in sign of the volumetric strain evolution, which coincides with a peak of acoustic emissions. At this point, cracks get closer to each other, which makes their propagation unstable. Finally, macroscopic failure happens when cracks coalesce, forming a fault plane, with subsequent sliding along its plane. Until that fault slip, dilatancy is a key mechanism controlling the failure in intact brittle rocks.

While this analysis details the general mechanical behavior of a crystalline rock, it does not explain how cracks induce anisotropy, and affect seismic velocities and attenuation. Indeed, cracks can cause the seismic waves to scatter and change direction, plus they provide pathways for seismic energy to directly escape from the rock [Lockner *et al.*, 1977; Paglialunga *et al.*, 2021]. So, cracks oriented perpendicular to the direction of the wave will cause more attenuation and seismic reduction than those that are oriented parallel to the wave. These principles allow observing two conflicting mechanisms during the experiments: i) when a rock is under pressure, pressure causes the cracks to close or partially close, delaying the onset of nucleation of new cracks. This causes an increase of velocities, a reduction of attenuation, and a decrease in crack densities. Similarly, under the first stage of applied vertical loading, horizontal cracks close. Less crack development is also observed at higher confining pressures. ii) In opposition, new cracks will develop after a critical stress threshold. These new cracks will grow parallel to the main principal stress and consequently affect elastic properties. Perpendicularly to these new cracks, velocities are reduced, and attenuation is increased. However, close to complete failure of the samples, a late horizontal opening of cracks occurs. Wing cracks coalesce and interact with shear cracks, which induce an average orientation of the cracks at 30° with respect to the axial stress. The ratio v_p/v_s (90°) that is used to predict failure [Gupta, 1973] reach also its largest values at this stage of the experiments.

Seismic attenuation demonstrates a strong correlation with changes in the mechanical behavior leading to failure (Figure 4c). The energy impulse of a seismic wave passing through a slightly opened crack is expected to be sufficient for its closure, resulting in energy loss and increased attenuation [Walsh, 1966; Lockner *et al.*, 1977]. Conversely, when the crack is fully closed, no changes in attenuation are observed. In our study, we employed the monitoring of P-wave amplitude as a proxy for attenuation, where a decrease in P-wave amplitude corresponds to an increase in attenuation. At $\sigma_3 > 24$ MPa, a subtle increase in amplitude was observed, indicating the complete closure of the majority of cracks, thereby hindering the aforementioned dissipative mechanism. Of particular interest, an increase in amplitude was observed for cracks oriented at a 35° angle under lower confining pressures, coinciding with the sample's failure. This suggests that, at $\sigma_3 < 24$ MPa, diagonally-oriented cracks maintain a state between complete closure and full openness, allowing for frictional dissipation induced by seismic wave pulses. Similar observations have been reported in previous studies [Lockner *et al.*, 1977]. However, a detailed analysis of these variations is challenging, as they potentially overlap with the growth of

487 new cracks, which produce the opposite effect. Nevertheless, these suggest that attenuation could
488 be a tool to monitor the proximity to failure if recorded from different directions.

489 5.2 Dissipation of the energy during the failure of brittle rocks

490 Here, we examine the validity of the wing crack model by analyzing the energy budget as-
491 sociated with inelastic mechanisms leading to the failure of intact brittle rocks. By comparing
492 the dissipated energies calculated using the experiments and the model, we assess its accuracy
493 and identify the micromechanisms contributing to brittle failure.

494 During the experiments, the inelastic energy is directly obtained from strain and stress mea-
495 surements, while the model also allows computing it from the variation of elastic wave speeds
496 during the experiments.

497 From strain measurements, the energy dissipated inelastically per sample is simply: $w_d =$
498 $w_{\text{tot}} - w_e$, with w_{tot} the total strain energy and w_e the elastic strain energy; these energies can
499 be directly computed from mechanical data as followed.

500 The total strain energy per sample is by definition:

$$w_{\text{tot}} = \int \sigma_{ij} d\varepsilon_{ij} = \int \sigma_1 d\varepsilon_1 + 2 \int \sigma_3 d\varepsilon_3 \quad (23)$$

501 because $\sigma_{ij} = 0$, $\forall i \neq j$ in the eigenspace formed by the principal stresses σ_1, σ_2 and σ_3 (no
502 shear stress is applied). The elastic strain energy is similarly obtained and as elastic strains are
503 by definition linear, it simplifies to:

$$w_e = \frac{1}{2E_0} (\sigma_1^2 - \sigma_1\sigma_3 + 2\nu_0(\sigma_3^2 - \sigma_1\sigma_3)) \quad (24)$$

504 The inelastic strain energy of the sample w_d is scaled by the size of the initial sample to provide
505 a comparable value: $W_d = hw_d$ [J/m²]. The computation of W_d requires an array of strain mea-
506 surements in reason of possible strain concentrations due to heterogeneous and non-linear rock
507 behavior.

508 The loss of elastic wave speeds is linked to a loss of stiffness and the propagation of cracks,
509 hence energy dissipated in crack propagation. Thus, the evolution of seismic velocities can be
510 used to compute the stiffness loss and then, the crack strain energy. Let the crack strain be ε^c .
511 The crack strain energy per sample is:

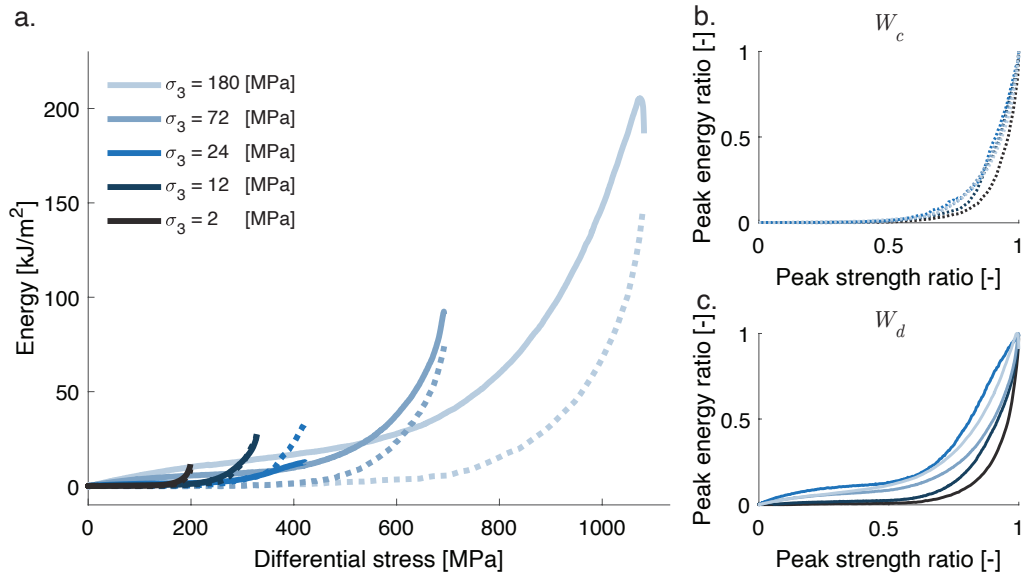
$$w_c = \int \sigma_{ij} d\varepsilon_{ij}^c \approx \frac{1}{2} \sigma_{ij} \varepsilon_{ij}^c \quad (25)$$

512 and $\varepsilon_{ij}^c = \Delta S_{ijkl} \sigma_{kl}$. This change of compliance is a function of the crack density tensor com-
513 ponents [Sayers and Kachanov, 1995]. Accounting a damage zone $w = 20$ [mm] as observed
514 by *Aben et al.* [2020] during failure thanks to localization of acoustic emissions and tomogra-
515 phy imaging, the energy becomes $W_c \approx \frac{1}{2} w \sigma_{ij} \Delta S_{ijkl} \sigma_{kl}$ [J/m²] [*Aben et al.*, 2019].

516 W_c and W_d exhibit similar trends and increase as the confining pressure rises (Figure 8a).
517 These inelastic energies increase despite the creation of similar quantities of new crack surfaces
518 (Figure 4b). Consequently, another pressure-sensitive process governs their behavior. Dilatancy,
519 which is associated with crack opening and accurately captured by the model due to the simi-
520 larity between W_c and W_d , emerges as the primary candidate. Consequently, in laboratory set-
521 tings, variations in elastic wave speeds can be used to measure the dissipation of inelastic ener-
522 gies prior to failure.

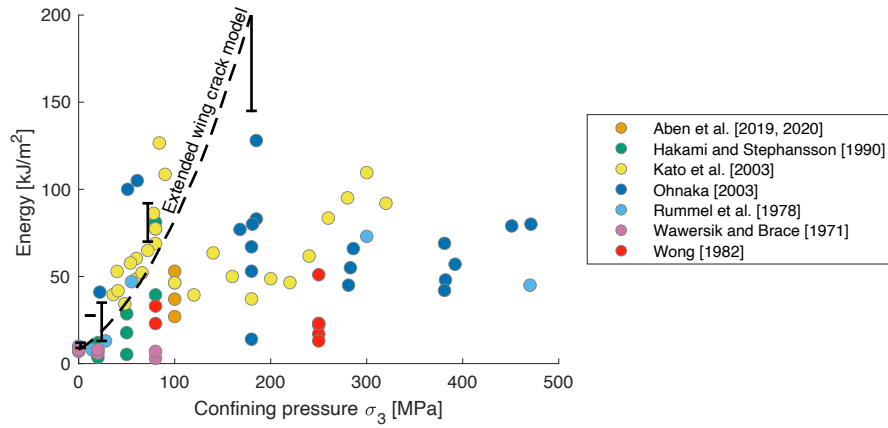
523 Despite the general similarity, W_c is slightly lower than W_d . Indeed, other inelastic dis-
524 sipation mechanisms that are not influencing velocity measurements or are not taken into account
525 in the computation of W_c may take place. Intracrystalline plasticity and diffusive mass transfer
526 are unlikely to occur under these conditions, but friction on crack surfaces may take place [*David*
527 *et al.*, 2020b; *Brantut and Petit*, 2022], as suggested by the attenuation measurements. Clues of

528 shearing can be seen when comparing in detail the energy dissipation in Figure 8b,c: with increas-
 529 ing confining pressure, W_d displays an initial plateau during loading, coming from shearing on
 530 cracks. Kachanov's theory does not consider inelastic shearing, so this plateau is absent for W_c .
 531 This results in shearing being observable in the larger mismatch between W_c and W_d at higher
 532 confining pressures. In short, the attenuation analysis and the energy budget both indicate there
 533 is negligible friction on crack faces below $\sigma_3 = 24$ [MPa].



534 **Figure 8.** a. Evolution of crack opening energies in function of differential stress according to two meth-
 535 ods. Dashed lines according to the stiffness loss associated with the evolution of seismic velocities (W_c), and
 536 full lines according to the strains and stress directly measured (W_d). b. and c. represent the same but normal-
 537 ized results for W_c and W_d respectively.

538 The energy dissipated into inelastic processes prior to the failure of intact rocks is signifi-
 539 cant in our experiments. Specifically, W_c and W_d increase from 20 to 200 kJ/m² with increas-
 540 ing confining pressure. Our theoretical estimates suggest that this energy could further increase
 541 at higher confining pressure, corresponding to greater depths. Remarkably, W_c and W_d are com-
 542 parable to the values of breakdown work estimated during the failure of intact rocks, which refers to
 543 the energy dissipated during the weakening of faulting [Wong, 1982a; Rummel et al., 1978].
 544 These results show that almost the same amount of energy is lost during microcrack formation
 545 in the preliminary stage of fracture as during macroscopic failure of the specimen itself (Fig. 9).
 546 However, these comparisons hold only at the laboratory scale, where the damage zone prior to
 547 failure is roughly equivalent to the nucleation zone size. Scaling these findings to geological set-
 548 tings faces challenges due to the cyclic localization of the damage zone, leading to significant
 549 variations in its size [Kato and Ben-Zion, 2021]. Nevertheless, inelastic processes occurring dur-
 550 ing the nucleation of instability are expected to alter the nucleation processes and result in larger
 551 energetic ruptures [Brantut and Viesca, 2015]. Consequently, although the magnitude of these
 552 energies in real earthquakes remains unknown, the precursory inelastic energy, is expected to en-
 553 hance the length scale of nucleation processes and improve our chances of detection. Therefore,
 554 we infer that precursory signs of failure may be observable through variations in elastic wave speeds,
 555 particularly at depth.



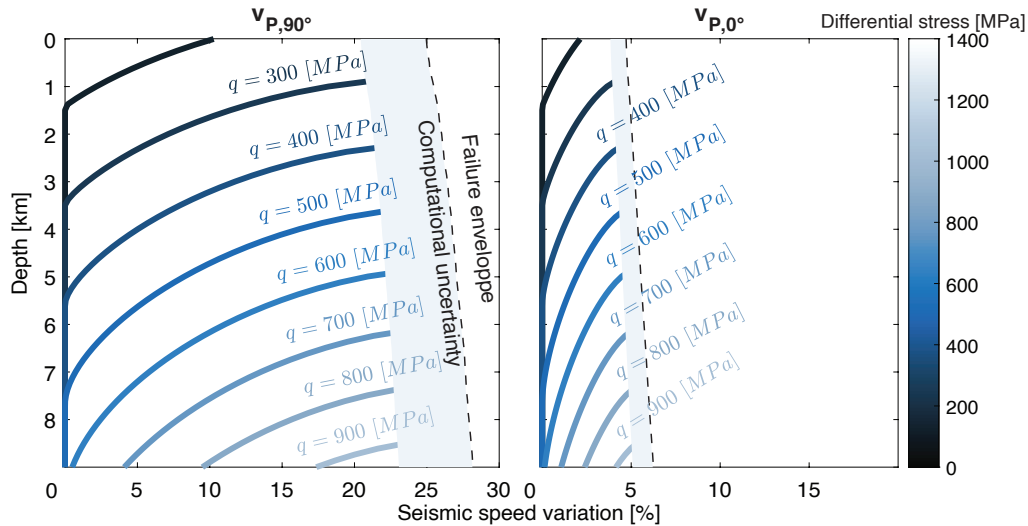
556 **Figure 9.** Comparisons between inelastic energy dissipated toward failure and during failure of granitoids
 557 in function of confining pressure. All circle points represent the breakdown work during macroscopic failure
 558 and slip on a main fault from Wawersik and Brace [1971]; Rummel et al. [1978]; Wong [1982b]; Hakami
 559 and Stephansson [1990]; Ohnaka [2003]; Kato et al. [2003]; Aben et al. [2019, 2020]. In black, our energy
 560 estimations before failure; the error bars show our measurements bounded by W_c and W_d . The dashed line
 561 represents modelled prediction according to the loss of seismic velocities.

5.3 Extrapolation to the Upper Crust

562
 563 So, how are expected to evolve the elastic velocities in the earth crust prior to brutal fail-
 564 ure? Indeed, the model and the experiments show a reduction of seismic velocities after a criti-
 565 cal stress. As the model fits the velocity variations, it can be used to predict these losses for any
 566 given loading. The practicality of possible applications is demonstrated with a theoretical exam-
 567 ple where *in-situ* stresses are estimated based on elastic wave speed variations.

568 Let us consider an imaginary case with its principal effective stresses: $\sigma_3 = (\rho - \rho_w)gz$,
 569 with z the depth in the crust, ρ the rock density, ρ_w the water density, and g the gravitational force.
 570 σ_1 is the main stress acting horizontally (the differential stress is $q = \sigma_1 - \sigma_3$). For simplifi-
 571 cation purposes, let $\sigma_2 = \sigma_3$. As the model predicts the elastic wave speed variations for σ_1 ,
 572 Figure 10 is obtained. Elastic wave velocities are dependent on the Young modulus, and this pa-
 573 rameter was estimated with the following empirical law (fitting our five Young Modulus mea-
 574 surements and velocity measurements of the initial pressure increase of WG3): $E_0(\sigma_3) = 3.75 \log(\sigma_3 +$
 575 $1) + 53.5$ [GPa]. The cracks open parallel to the principal stress, so the ray path angle is criti-
 576 cal. With this graph, we can estimate the differential stress in relation to depth, ray path angle,
 577 and seismic speed reduction. Time-dependent effects such as subcritical crack growth or crack
 578 relaxation are not taken into account for this example.

582 This simple example shows that velocity reductions are principally perpendicularly to the
 583 main principal stress and at high differential stress. For instance, if we observe a 10% v_p reduc-
 584 tion at a 5 [km] depth, it is associated with an applied 500 [MPa] differential stress. As crack in-
 585 duced anisotropy is a nearly reversible process, this differential stress inversion remains accu-
 586 rate despite the loading history for brittle rocks [Bonnelye et al., 2017; Passelègue et al., 2018].
 587 Another factor to take into account in this interpretation is the size of the zone where new cracks
 588 are created. In geological settings, damage concentrates around faults in a so-called damage zone
 589 and its width varies with depth and tectonic settings. For instance, it can be kilometeric at the sur-
 590 face on reverse faults, to decametric at depth [Caine et al., 1996; Mitchell and Faulkner, 2009].
 591 This width is crucial, as seismic variations will appear relatively smaller proportionally to the scale
 592 of observation. Therefore, variations at depth might get unnoticed despite important variations.



579 **Figure 10.** A wing crack model application to estimate elastic wave speed reductions in function of the
 580 principal stress σ_1 , the angle between the ray path and σ_1 , and the depth. For this example, σ_2 and σ_3 are
 581 computed in function of depth from a hydrostatic estimation of the stress.

593 In various studies monitoring the evolution of velocities in fault zones, velocity drops were
 594 observed after earthquakes followed by a post-seismic relaxation [Brenguier *et al.*, 2008]. The
 595 coseismic velocity reduction might be caused by coseismic damage and reopening of existing
 596 cracks. At first glance, these observations seem to contradict laboratory experiments and the model.
 597 However, by using the same method as Brenguier, it has also been observed that the earthquake
 598 velocity drop is reduced with depth [Hobiger *et al.*, 2012], which suggests that the drop is mainly
 599 caused by the reopening of existing cracks at low stresses [Meyer *et al.*, 2021; Paglialunga *et al.*,
 600 2021]. As explained by fracture energy comparisons, our experiments are conducted on intact
 601 rocks, whose properties are different from fault zones or cracked solids. The strength of fault zones
 602 is generally low, so the crack nucleation process does not occur in stick-slip earthquakes at low
 603 depth [Brace *et al.*, 1966], meaning the wing crack model cannot be applied in these conditions.
 604 The model might still be applicable for intact rocks or at depth where cracks are healing.

605 6 Summary

- 606 1. Experiments on intact Westerly granite documented the evolution of crack nucleation with
 607 detailed elastic wave velocities and attenuation measurements. From damage inversion,
 608 we observed that cracks grow parallel to the principal stress, except close to failure at low
 609 confining pressures. Dilatancy is the main phenomenon controlling failure of intact brittle
 610 rocks. Shearing on cracks and defects only plays a critical role at high confining pres-
 611 sures (> 24 [MPa]).
- 612 2. The micromechanical wing crack model of Ashby and Sammis [1990], modified by Desh-
 613 pande and Evans [2008], has been extended and linked to the cracked solid theory of Say-
 614 ers and Kachanov [1995] with simple considerations on the cracks geometry. The model
 615 predicted the evolution of elastic wave velocities during loading of intact granite.
- 616 3. A budget of the energy dissipated to open cracks according to the mechanical results and
 617 the seismic velocities showed a compatibility with the use of the wing crack model with
 618 Kachanov's theory. Comparisons with literature estimates show that the inelastic energy
 619 dissipated prior to failure is of the same order as the breakdown work.

620 4. Therefore, the use of this model for geophysics applications is conceivable, but only at
 621 depth or for intact rocks. In-situ stresses and crack-induced anisotropy might be estimated
 622 with seismic velocities.

623 Acknowledgments

624 M.V. acknowledges support by the European Research Council Starting Grant project 757290-
 625 BEFINE.

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